# Electrical failure, Boeing 767, June 19, 1996

Micro-summary: Dual generator failure results in a trip back to the gate on this Boeing 767.

Event Date: 1996-06-19 at 2322 MDT

Investigative Body: National Transportation Safety Board (NTSB), USA

Investigative Body's Web Site: http://www.ntsb.gov/

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National Transportation Safety Board
FACTUAL REPORT
AVIATION

NTSB ID: SEA96IA131 Aircraft Registration Number: N136DL

Occurrence Date: 06/19/1996 Most Critical Injury: None

Occurrence Type: Incident Investigated By: NTSB

Location/Time

Airport Proximity: Off Airport/Airstrip	Distance From	n Landing Facility:		Direction Fro	m Airport:
SALT LAKE CITY	UT	84116	2322	MDT	
Nearest City/Place	State	Zip Code	Local Time	Time Zone	

Aircraft Information Summary

Aircraft Manufacturer	Model/Series	Type of Aircraft
Boeing	767-332	Airplane

Sightseeing Flight: No Air Medical Transport Flight: No

#### Narrative

Brief narrative statement of facts, conditions and circumstances pertinent to the accident/incident:

HISTORY OF FLIGHT

On June 19, 1996, at 2322 hours mountain daylight time, a Boeing 767-332, N136DL, being operated by Delta Airlines as flight 229, sustained minor damage when the number two (right) engine integrated drive generator (IDG) caught fire during a climb, approximately 80 nautical miles northwest of Salt Lake City, Utah. The Captain, First Officer, six cabin crew, and 178 passengers were uninjured. Visual meteorological conditions prevailed and an IFR flight plan was in effect. The flight, which was a regularly scheduled passenger flight operated under 14CFR121, departed Salt Lake City International airport at 2308 destined for Portland, Oregon.

During taxi for takeoff the crew observed both the "R GEN OFF" and "R GEN DRIVE" EICAS (engine indication and crew alerting system) messages and returned to the gate for corrective action. The captain reported that he "notified maintenance control and informed them that the aircraft was returning to the gate and informed them of the EICAS messages" and that the "after landing checklist was completed as well as the "generator off" abnormal procedure" (refer to ATTACHMENT I). He further reported that he "was turning the aircraft into the gate area, therefore further checklist procedures were discontinued pending examination of the aircraft by maintenance personnel." The aircraft's engines were then shut down.

According to the Captain, a mechanic was briefed on the problem and departed the flight deck with the aircraft log and minimum equipment list (MEL). Shortly thereafter, the mechanic returned with the aircraft log showing the discrepancy as having been signed off. The captain reported that he discussed the maintenance actions taken and was told by the mechanic that "the IDG is disconnected." The crew anticipated observing the generator drive light and right generator off light for the duration of the flight to Portland, and would be using the auxiliary power unit (APU) generator as a replacement for the right generator.

The mechanic reported that upon entering the cockpit he "found the right generator field light ON with (the) associated IDG drive light illuminated." He further reported that he "found the right generator drive disconnect switch released (and) assumed (the) crew had disconnected (the) IDG." The aircraft was subsequently dispatched in accordance with the MEL (refer to attached statement and ATTACHMENT II).

After an uneventful taxi, takeoff and departure, and while crossing flight level (FL) 250, the crew observed a "R STARTER CUTOUT" EICAS message and the overhead right starter valve light illuminated intermittently. The "Starter Cutout" procedure was accomplished following which a flight attendant reported that a passenger in the aft cabin, right side, had observed sparks coming from the right engine. This was confirmed by the First Officer's visual inspection.

The First Officer then returned to the flight deck and the crew initiated a return to Salt Lake

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Narrative (Continued)

City during which the right engine fire warning light illuminated. The right engine was shut down and both fuselage mounted fire bottles were exhausted into the right engine with the fire warning light ceasing shortly after the second bottle was fired. An uneventful single engine emergency landing was accomplished at 2343 on runway 16L at Salt Lake City, and the aircraft rolled to a stop in the vicinity of the taxiway H-3 exit (refer to DIAGRAM I). The Captain established radio communications with the aircraft rescue and fire fighting (ARFF) team. The left engine was shut down shortly after landing with the aircraft remaining on APU power. According to ARFF records maintained by Salt Lake City airport, the fire was contained approximately 7 minutes after the landing. The left engine was subsequently restarted and the aircraft was taxied to gate D-3 under ARFF escort after which the passengers were deplaned via the jetramp.

#### POST INCIDENT EXAMINATION AND INFORMATION

Onsite examination of the right engine by Federal Aviation Administration (FAA) inspectors from the Salt Lake City Flight Standards District Office revealed that the IDG had separated from the accessory section of the right engine. The remnants of the IDG, including what remained of its magnesium housing were removed. There was evidence of fire damage within the accessory section of the engine concentrated in the vicinity of the IDG and hydro-mechanical fuel control unit. A hole was observed burned through the starter gearcase adjacent to the IDG and the start valve actuator was burned away (refer to DIAGRAM II). No significant damage was noted within the engine core area and the titanium heat shield which separates the accessory section from the core had only a single hole burned through the aft vertical face aft of the IDG location. A single hole was observed burned through the engine core cowling and in alignment with the previously described heat shield hole.

Additionally, FAA inspectors confirmed that the shear wire on the cockpit stationed IDG disconnect switch was intact.

## AIRCRAFT INFORMATION

The fire detection element which activates the cockpit fire warning annunciator does not pass near the IDG but is situated near the hydro-mechanical fuel control unit. The aircraft's General Electric CF6-80A engines are equipped with a titanium heat shield which separates the accessory section from the engine core. Engines not equipped with this heat shield (Pratt & Whitney and Rolls Royce) are equipped with a second fire detection element within the accessory section.

There is only one procedure (exclusive of removal) for physically disconnecting an IDG from its associated engine. The procedure entails breaking the shear wire which holds the clear plastic guard over the disconnect switch. This allows the hinged plastic guard to be rotated away from the square, push-button disconnect switch. The switch can then be depressed (refer to DIAGRAM III). When the switch is depressed it initiates the physical disconnection of the IDG from the engine, thus preventing rotation of the IDG armature. This procedure may be accomplished by the flight crew during flight or ground taxi operations by executing the "GENERATOR DRIVE" abnormal checklist with the engines running (ATTACHMENT I), or by maintenance as part of the MEL dispatch procedure for an inoperative left or right IDG (refer to ATTACHMENT II, pages one and two). This latter procedure (ATTACHMENT II) was a procedure developed by Delta Airlines from the Boeing 767 Dispatch Deviations Guide. In order for an IDG to be properly disconnected during this maintenance procedure without damaging the IDG's "disconnect dog teeth," it is necessary to start and run the associated engine at or above idle power before carrying out the disconnect procedure (item "C" of ATTACHMENT II, page one).

# TESTS AND RESEARCH

A readout of digital data downloaded from the aircraft revealed that original IDG light and EICAS messages observed by the flight crew on initial taxi were related to low oil pressure within the

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Narrative	(Continued)			

IDG, and that the fire warning light observed by the crew during climbout extinguished shortly after the second fire bottle was exhausted into the right engine.

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AVIATION Occurrer					rrence Type: Incident								
Landing Facility/Approach Information													
Airport Name			Airpor	t ID:	Airport Eleva	tion	Run	way Used	Runwa	ay Lengt	h	Runw	ay Width
SALT LAKE CITY INTL			SLC		4227 Ft	. MSL	161	_	1200	3		150	
Runway Surface Type: Asphalt													
Runway Surface Condition: Dry													
Type Instrument Approach:													
VFR Approach/Landing: Straight-in													
Aircraft Information													
Aircraft Manufacturer				Model/9 767-3						Serial 2514	Numbe s	r	
Boeing  Airworthiness Certificate(s): Trans				101-3						2014			
An worthiness Certificate(s). Trans	sport												
Landing Gear Type: Retractable - Tricycle													
Homebuilt Aircraft? No	Number of Seats: 2	264	c	Certified	Max Gross W	/t.		345000		Number of Engine		gines:	2
				Engine Manufacturer: Model/Series: CF6-80A2								d Power: 70 LBS	
- Aircraft Inspection Information													
Type of Last Inspection			Date of Last Inspection Time Sir			ince Last Inspection			Airfram	ne Tot	tal Time		
Continuous Airworthiness			06/1	06/1996				45 Hours					Hours
- Emergency Locator Transmitter (	ELT) Information												
ELT Installed? Yes	ELT Operate	ed? No	)			ELT	Aided i	n Locating A	ccident S	Site?			
Owner/Operator Information													
Registered Aircraft Owner			S	treet A	ddress RODNE`	Y SQ	UARE I	NORTH					
WILMINGTON TRUST COMP	ANY		Ci	City							State		Zip Code
			WILMINGTON DE 19890 Street Address								19890		
Operator of Aircraft				11001710		IELD	-ATLAI	NTA INTNL	ARPT				
DELTA AIR LINES			City							State		Zip Code	
Operator Does Business As:					ATLANTA GA 30320  Operator Designator Code: DALA							30320	
- Type of U.S. Certificate(s) Held:													
Air Carrier Operating Certificate(s)	: Flag Carrier/Dom	nestic											
					T								
Operating Certificate:					Operator (	Sertific	cate:						
Regulation Flight Conducted Unde					0.1								
Type of Flight Operation Conducted	d: Scheduled; Dor	nestic;	; Pass	enger	Only								
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Occurrence Date: 06/19/1996

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AVIATION Occurrence Type: Incident						ident							
First Pilot Information													
Name		City	State			С	ate of Birth	Age					
On File		On File	le On File On File					On File	53				
Sex: M	Seat Occupied	Left	Prin	ncipal Profes	sion: Civilia	n Pilot			Ce	rtificate	Numbe	er: On File	
Certificate(s): Airline Transport; Commercial; Flight Engineer													
Airplane Rating(s): Multi-engine Land; Single-engine Land													
Rotorcraft/Glider/LTA: None													
Instrument	Rating(s): Airpl	ane											
Instructor F	Rating(s): None	e											
Type Rating	g/Endorsement fo	or Accident/In	cident Aircra	ft? Yes			Current I	Bienn	ial Flight F	Review?			
Medical Ce	rt.: Class 1	Medica	al Cert. Status	: Valid Me	dicalw/ wa	aivers/lin	٦.		Date of La	ast Medi	ical Ex	am: 03/1996	
		·											
- Flight Tim	e Matrix	All A/C	This Make and Model	Airplane Single Engine	Airplane Mult-Engine	Night	Actua	Instrument Simulat		Rot	torcraft	Glider	Lighter Than Air
Total Time		19500	962										
Pilot In Con	nmand(PIC)												
Instructor													
Last 90 Day	/S	166	166										
Last 30 Day	/S												
Last 24 Ho	urs					<u> </u>							
Seatbelt Us	sed? Yes	Shou	lder Harness	Used? Yes		Т	oxicology P	erforn	ned? No		Sed	cond Pilot? Ye	es
	n/Itinerary pht Plan Filed: IF												
Departure F		K				Т.	State	Airport Identifier Departu			uro Timo	Time Zone	
	Accident/Incide	nt Location				,	State	SLC			рерапі 2255	ure Time	MDT
Destination							Diata .	Λ:					
PORTLAND							State Airport Identifier OR PDX						
Type of Clearance: IFR													
Type of Airspace: Class A													
Weather Information													
Source of Briefing:  Company													
Method of	Briefing:												
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Weather Information													
WOF ID	Observation Time	Time Zone	WOF	Elevatio	vation WOF Distance From Accident Site						Direction From Accident Site		
SLC	2352	MDT	4	227 Ft. N	√ISL				0 NM			0 De	eg. Mag.
Sky/Lowes	st Cloud Condition: Scat	tered				15	5000 Ft. AG	GL.	Condition o	of Ligh	nt: Night/Dark		
Lowest Ce	iling: None			0 Ft. A	\GL	Visibi	ility:	10	SM	Altii	meter:	29.00	"Hg
Temperatu	ıre: 17 °C	Dew Point:		8 °C	Wind	Direction:	120			Der	nsity Altitude:		Ft.
Wind Spee	ed: 3	Gusts:			Weatl	her Condti	ions at Acci	dent S	ite: Visual C	Condi	itions		
Visibility (R	RVR): 0 Ft.	Visibility	(RVV)	0	SM	Intensity	y of Precipit	ation:	Unknown				
Restriction	s to Visibility: None												
Type of Pre	Type of Precipitation: None												
Accident	Information												
Aircraft Dar	mage: Minor		Airo	craft Fire:	In-fliç	jht			Aircraft Exp	losio	n None		
Classificati	on: U.S. Registered/U	.S. Soil											
- Injury Sur	mmary Matrix	Fatal	Serious	Minor		None	TOTAL						
First Pil	lot					1	1	]					
Second	d Pilot					1	1						
Student	t Pilot							1					
Flight Ir	nstructor							1					
Check F	Pilot							1					
Flight E	Engineer							]					
Cabin A	Attendants					6	6	]					
Other C	rew							]					
Passen	igers					178	178						
- TOTAL A	ABOARD -				T	186	186	]					
Other G	Fround	0	0	,	0		0	]					
- GRAND	TOTAL -	0	0	,	0	186	186						

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FACTUAL REPORT  AVIATION	Occurrence Date: 06/19/1996								
AVIATION	Occurrence Type: Incident								
Administrative Information									
Investigator-In-Charge (IIC)									
STEVEN A. MCCREARY									
Additional Persons Participating in This Accident/Incident	ent Investigation:								